

X1 - 120

Cosmetic Matrix Epoxy Prepregs

TECHNICAL DATA SHEET

X1 MATRIX SERIES is a thermosetting epoxy matrix series with high cosmetic results¹ and is available in several product variants. X1 matrix exhibit high mechanical properties and, properly postcured, can be used at continuous operating temperatures up to 80°C. Higher not continuous operating temperatures can be supported.

PRODUCT VARIANTS

X1-120: Solvent version, unpigmented

X1-120HM: Hotmelt version

SHELF LIFE



OUT LIFE 28 days @ 21 °C



STORAGE LIFE 12 months @ -18 °C

TYPICAL APPLICATIONS





FEATURES

GREAT COSMETIC PROPERTIES

GOOD MECHANICAL PROPERTIES

Note: All technical information contained in this document are given in good faith and are based on tests believed to be reliable, but their accuracy and completeness are not guaranteed. They do not constitute an offer to any person and shall not be deemed to form the basis of any contract. Accordingly, the user shall determine the suitability of the products for their intended use prior to purchase and shall assume all risk and liability in connection therewith. The information contained herein is under constant review and liable to be modified. All products are sold subject to Microtex Composites Srl terms and conditions of sale. Copyright 2020 - Microtex Composites Srl. All rights reserved worldwide. All trademarks or registered trademarks are the property of their respective owners.

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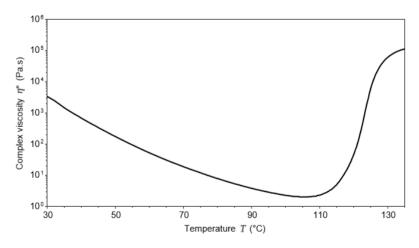
Where the intended end application is for a cosmetic product, customers are advised to consult a Microtex Composites sales representative for specific advice on fibre selection when placing an order for material.



MATRIX PROPERTIES

Cured resin density @ RT: (average value) 1.20 g/cm³.

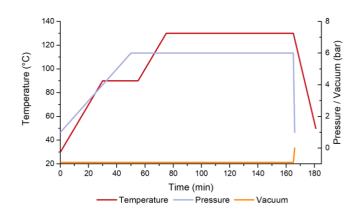
Resin viscosity: ramp rate = $2 \, ^{\circ}$ C/min, v = $10 \, \text{rad/sec}$.



CURING CONDITIONS

Preliminary Note: The matrix rheology, reaction times and final component surface quality are all affected by the chosen heat up rates. Heating rates are generally related to components size (large and thick components require slow heating rates). The heat up rate selected should avoid large temperature differentials between the component, tool and the heat source. For certain configurations and for most large components, an intermediate dwell can also be introduced into the cure cycle. It will guarantee even temperature distribution throughout the tooling and component. Good temperature control will provide consistent and improved resin flow characteristics during cure. To ensure that the matrix stability is fully developed, no polymerization residual should be present on the products.

C ption 1 - Au oclave Cure ^{2,3}			
Time	Temp.	Time	Pressure
(min)	(°C)	(min)	(bar)
0	30	0	1
30	90	10	2
55	90	30	4
75	130	50	6
165	130	165	6
181	50	166	1



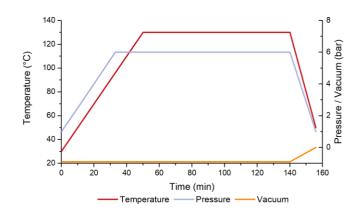
² Temperature must be measured by the lagging thermocouple attached to the part.

³ Vacuum bag pressure: 0.9 bar.



CURING CONDITIONS

C ption 2 - Autoclave Cure ^{4,5}			
Time	Temp.	Time	Pressure
(min)	(°C)	(min)	(bar)
0	30	0	1
50	130	10	2.5
90	130	20	4
130	130	33	6
140	130	140	6
156	50	156	1



ALTERNATIVE CURING CYCLES AND Tg's

X1-120			
Cure cycle	Tg (DSC) (°C)	Tg (DMA)*	Tg (DMA)*
		Onset (°C)	tanδ (°C)
8h @ 80°C	95÷100	-	-
6h @ 100°C	100÷105	-	-
60 min @ 120°C	110÷115		
60 min @ 130°C	115÷120	-	-
90 min @ 130°C	120÷125	115	130
Max Wet Tg ⁷	-	-	87

 $[\]hbox{* The reported DMA values are obtained from specimens consisting of light supports (\it{e.g.}~200~gsm~carbon~fabric).}$

 $^{^{4}}$ Temperature must be measured by the lagging thermocouple attached to the part.

⁵ Vacuum bag pressure: 0.9 bar.

 $^{^6}$ Wet conditioning: 14 days immersion in water @ 70 °C.



MECHANICAL PROPERTIES

X1-120 - 90 min @ 130°C, 6 bar		GG245T-427	GG290P-37 8	GG380T-42 ⁹	GG630T-3410
Property	Test Method	Value*			
0° Tensile strength [MPa]	-	707	1188	1062	974
0° Tensile modulus [GPa]		66	74	62	-
0° Tensile Poisson ratio	- ASTM D3039	0.05	-	0.05	-
90° Tensile strength [MPa]	- ASTM D3039	670	1223	1035	815
90° Tensile modulus [GPa]	•	64	75	62	-
90° Tensile Poisson ratio	•	0.05	-	0.05	-
0° Compressive strength [MPa]		697	682	624	-
0° Compressive modulus [GPa]	- ASTM D6641	63	69	64	-
90° Compressive strength [MPa]	- ASTM D0041	675	613	582	-
90° Compressive modulus [GPa]	•	66	73	51	-
0° Compressive strength [MPa]	SACMA	-	-	-	577
90° Compressive strength [MPa]	SRM 1R-94	-	-	-	548
0° Interlaminar shear strength (ILSS) [MPa]	ASTM D2344	-	61	63	70
0° Flexural strength [MPa]		1030	-	1048	-
0° Flexural modulus [GPa]	ASTM D7264	83	-	91	-
90° Flexural strength [MPa]	(B METHOD)	975	-	978	-
90° Flexural modulus [GPa]	•	82	-	87	-
0° Flexural strength [MPa]		-	799	1103	868
0° Flexural modulus [GPa]	A COTTAL DOCC	-	53	66	-
90° Flexural strength [MPa]	- ASTM D790	-	829	-	-
90° Flexural modulus [GPa]	-	-	55	-	-
Mode I Strain Energy release Rate G1c [J/m²]	ASTM D5528 (MBT METHOD)	-	-	900-1000	724

^{*} Test conditions: room temperature, dry . Normalized values at 55% VF .

 $^{^7}$ Carbon fabric 245 gsm twill 2/2 3K TC, RC 42%.

⁸ Carbon fabric 290 gsm plain 24K MR60, RC 37%.

⁹ Carbon fabric 380 gsm twill 2/2 12K ZH, RC 42%.

 $^{^{10}}$ Carbon fabric 630 gsm twill 2/2 12K ZH, RC 34%.



OTHER PROPERTIES

Citotoxicity: ISO 10993-5 2009: Carbon Laminate - Thickness 1.2 mm¹¹

Result: test specimen did not show cytotoxicity effect (skin contact).

Fluid compatibility:

X1-120			
	Carbon Laminate ¹²	Carbon Laminate ¹³	
Fluid type	1 day Absorption @ RT	28 days Absorption @ RT	
Fuel Total WEC 2018	+0.013%	+0.056%	
Oil Elf Huile HTX 860	-0.089%	+0.569%	
Coolant Elf RF1-CO-02	+0.345%	+0.488%	

AVAILABILITY

X1-120 series prepregs are available in a wide range of reinforcing fabrics, including carbon, aramid, glass and special fabrics.

STORAGE CONDITIONS

This prepreg should be stored as received in a cool dry place or in a refrigerator.

After removal from refrigerated storage, prepreg should be allowed to reach room temperature before opening the polyethylene bag, thus preventing condensation (a full roll in its packaging can take more than 1 day).

CHARACTERISTIC MARK

Possible shelf-life reduction for prepregs of higher weights (>400 gsm).

PRECAUTIONS FOR USE

The usual precautions when handling uncured resins and fibrous materials should be observed, and a Safety Data Sheet is available for this product.

SDS Reference Codes: X1-120: SIS-402

 $^{^{11}}$ Various configurations of fibers and fabrics (no painting, glossy painting, matt painting).

¹² Carbon fabric 245 gsm twill 2/2 3K TC, RC 40%; cure cycle 90'@130°C (no painted).

¹³ Carbon fabric (glossy painted).



EXOTHERM RISK

This matrix system can undergo severe exothermic heat up during the curing process if incorrect procedures are followed. Great care must be taken to ensure that safe heating rates, dwell temperatures and lay-up/bagging procedures are properly executed, especially when molding solid laminates with more than 8 mm thickness. The risk of exotherm increases with lay-up thickness and increasing of temperature cure. It is strongly recommended that the user identifies a safe cure cycle through trials that are representative of all the relevant processing parameters. It is also important to recognize that the model or tool material and its thermal mass, combined with the insulating effect of breather/bagging materials can affect the risk of an exotherm.

Please contact our technical department for further information on the exotherm behavior of these systems.

	Exotherm Test 1	Exotherm Test 2
Material	GG600T 12K X1-120 - RC 36%	GG600T 12K X1-120 - RC 36%
Final laminate thickness	16 mm	16 mm
Mold	Aluminium	Aluminium
Autoclave Cure Vacuum bag pressure	0.9 bar	0.9 bar
Autoclave Cure Pressure	4 bar	4 bar
Townseature change	Ramp 2°/min 20 to 100 °C	Ramp 2°/min 20 to 110 °C
Temperature steps:	Isotherm 80' @ 100 °C	Isotherm 80' @ 110 °C
Maximum temperature recorded on the part	120 °C	130 °C